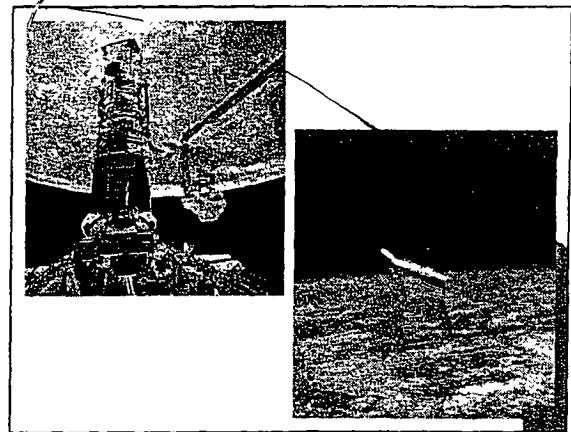
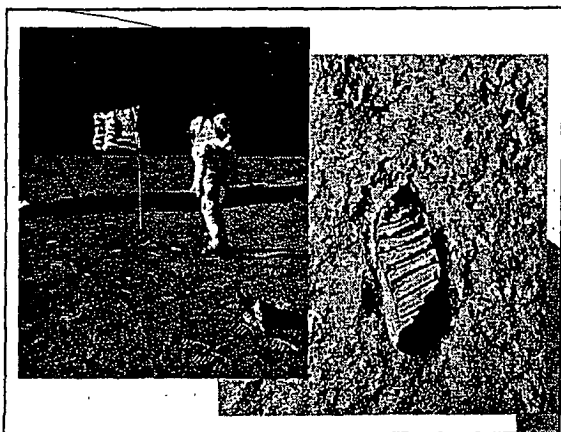
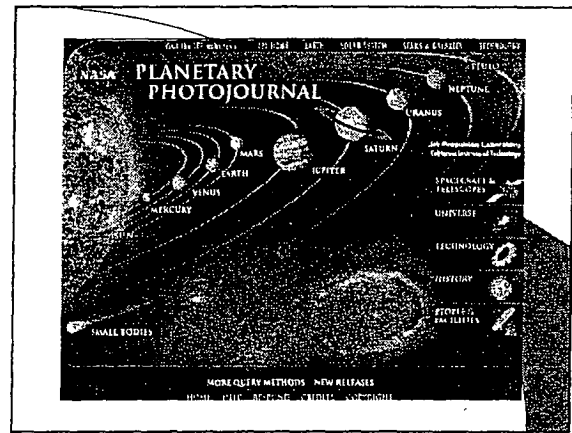
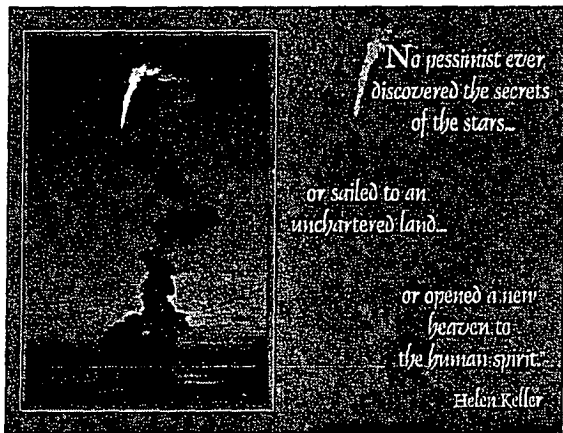
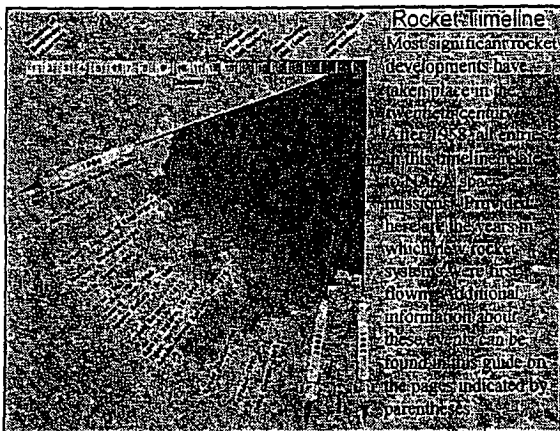
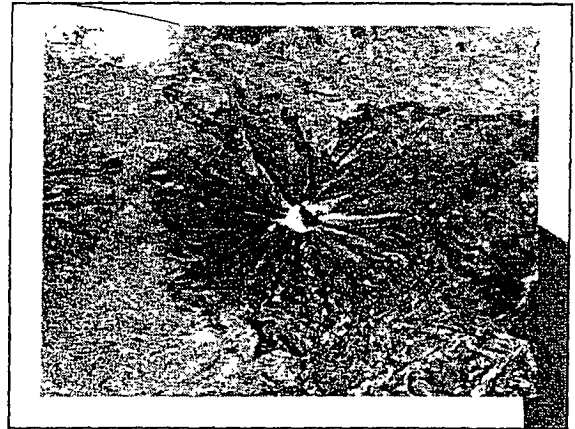
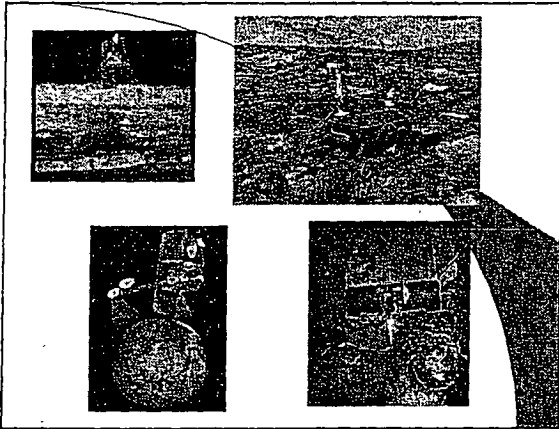
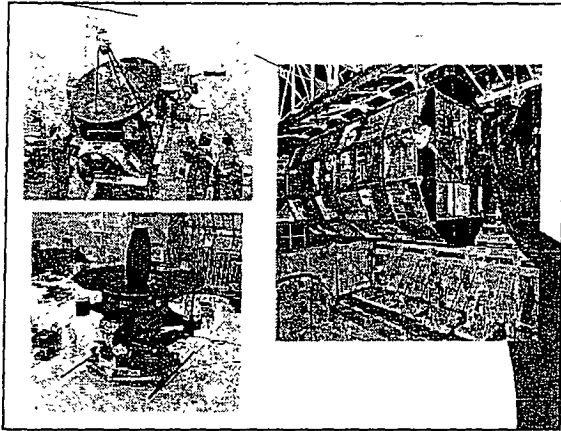


**Launch Pad Vibroacoustics  
Research at NASA / KSC**

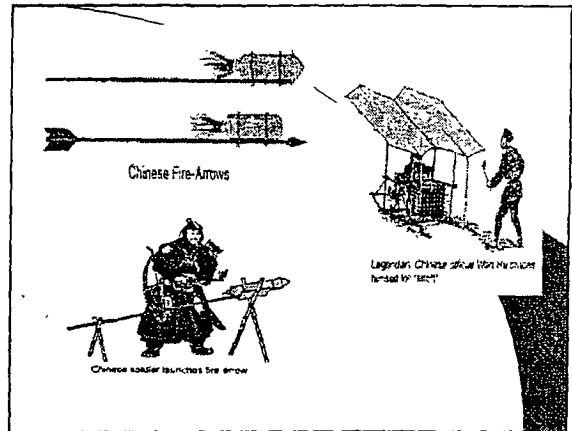
By:  
Ravi Margasahayam  
NASA Kennedy Space Center (KSC)  
Florida, USA

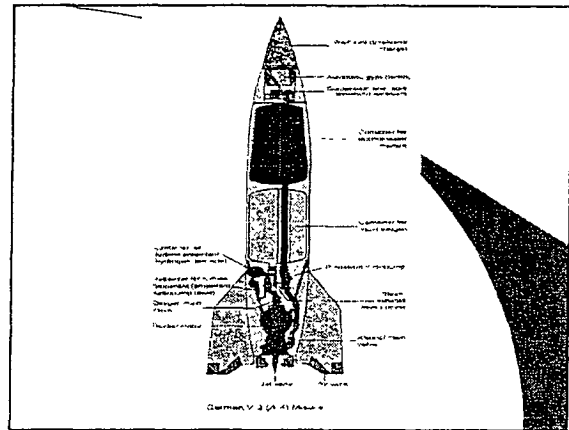
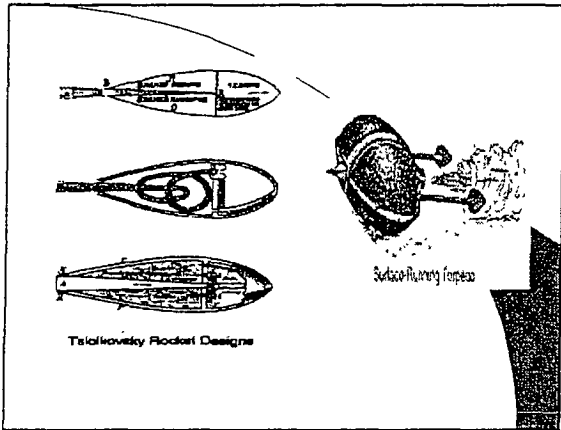
Presented At:  
Universidad Austral de Chile  
Valdivia, Chile  
December 12-13, 2002





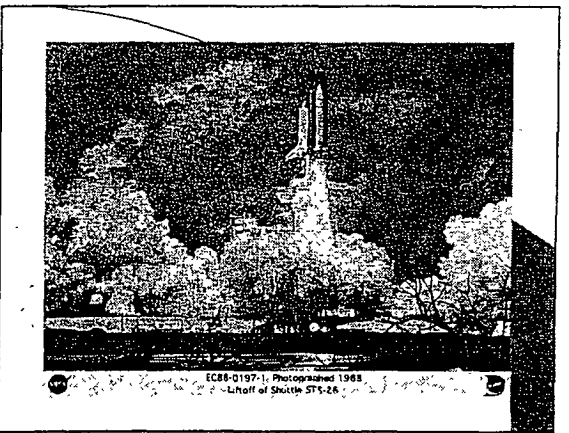
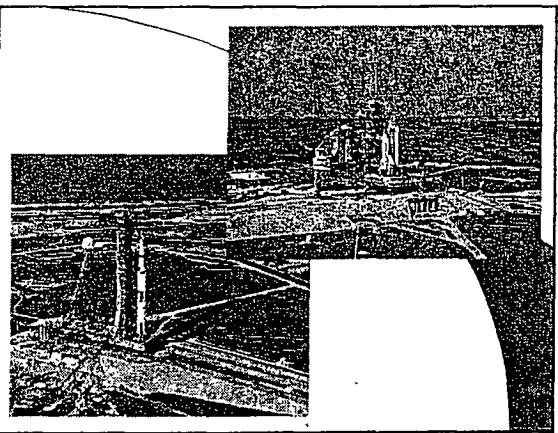
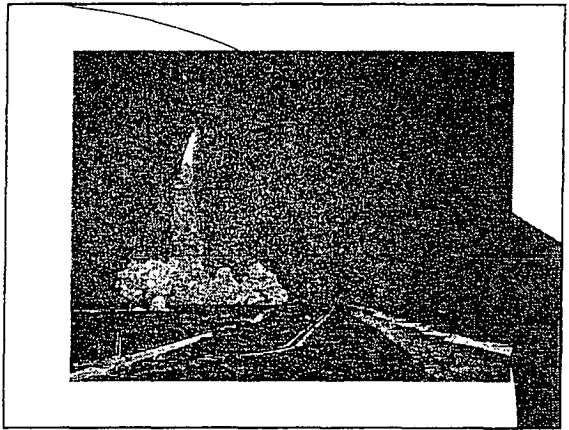
**Rocket Timeline:**  
 Most significant rocket developments have taken place in the 20th century. After 1945, all major milestones in this timeline are the result of the work of the United States, the Soviet Union, and China. The first successful launch of a rocket system was in 1926, and the first orbital launch was in 1957. The first human spaceflight was in 1961, and the first lunar landing was in 1969. The first Mars landing was in 1971, and the first Mars rover was launched in 1997. The first Mars orbiters were launched in 1999, and the first Mars lander was launched in 2001. The first Mars rover was launched in 2003, and the first Mars rover was launched in 2003. The first Mars rover was launched in 2003, and the first Mars rover was launched in 2003.

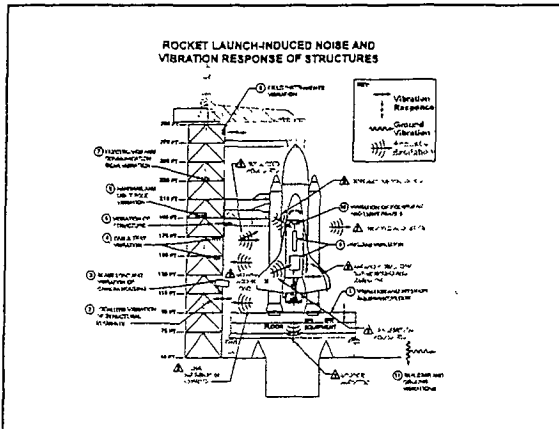




### Rocket Noise and KSC

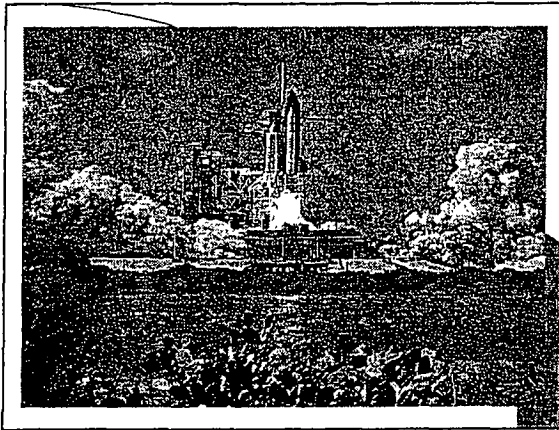
- KSC's role as a premier launch site dictates reliability and availability of pad and ground support equipment
- Structural vibration is consequent to launch acoustics, both air-borne and structure-borne
- Resonance is always present, due to wide band acoustic excitation
- Noise and Vibration measurements are key to launch readiness





## Characteristics of Noise

- Space Shuttle launch environment (plume pressures, acoustics, strains, vibration) are random, nonstationary, non-gaussian, and wideband
- Amplitude and frequency vary widely
- Significant variability exists between short duration and long duration data collection and processing thereof
- Large sample size needed (we have over 100 launches)

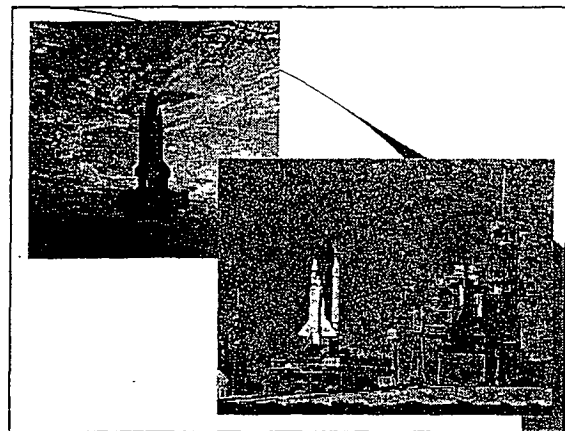


## Noise...Generation

- Space Shuttle noise: 180 dB ~ 2.8 psi
- Space Shuttle Lift-off:
  - Creates intense noise
  - Noise source is supersonic exhaust
  - Turbulent eddies created due to mixing of hot gases with ambient air...lead to shock
  - Shock strongest source of noise (see 5)

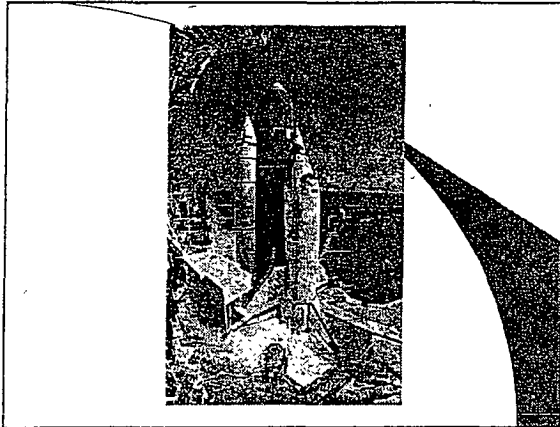
## Noise...Impact

- Vibroacoustics effect on the vehicle and its' valuable payload
- Effect on Astronauts inside vehicle
- Influence on pad and ground structures
- Impact on wildlife and ecosystem in close proximity of the pad
- Physiological and psychological effect on humans and surrounding community
- Sonic boom related issues









### Noise...SSME Ignition

- Loud roar and heavy vibrations in the cockpit ~ driving a car down a railroad track or sitting close to speakers at a rock concert
- SSME's produce close to a million pounds thrust; since rocket is held to the pad via bolt/nut arrangement
- Noise is LOUD and teeth-rattling

### Sounds...SRB Ignition

- Noise and vibration increase significantly; one cannot simulate such noise and shaking in the laboratory
- Vibrations and noise are not so bad that Astronauts cannot read instruments; noise does not prevent them from hearing fellow Astronauts (they use intercom for communication)

### Noise...in Space

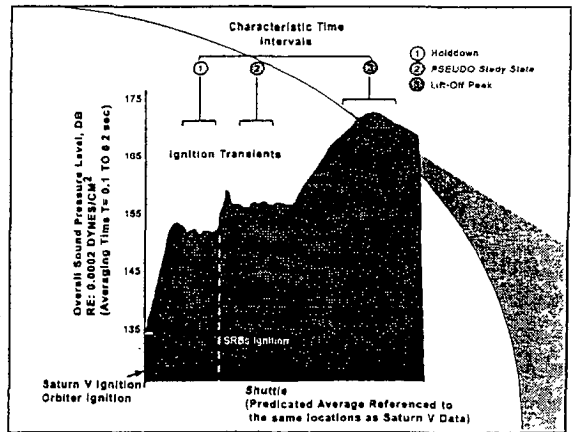
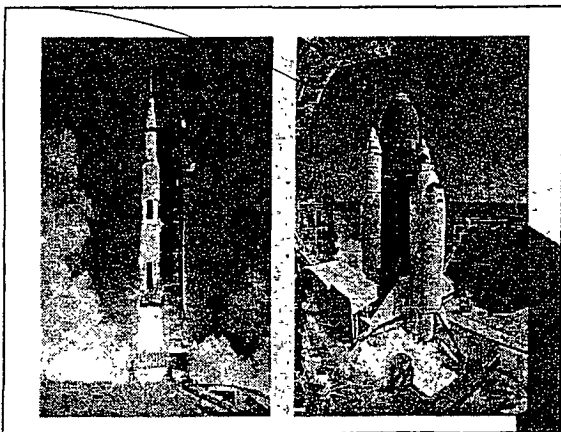
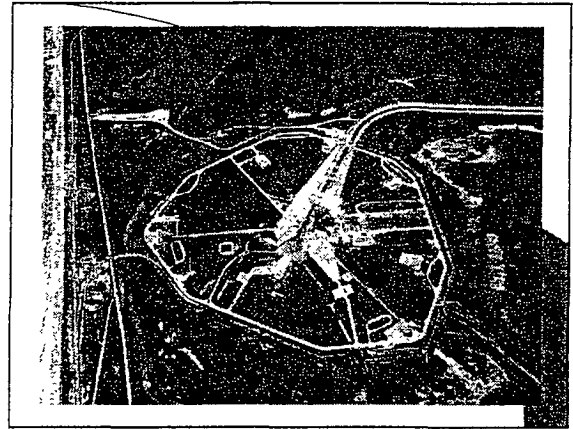
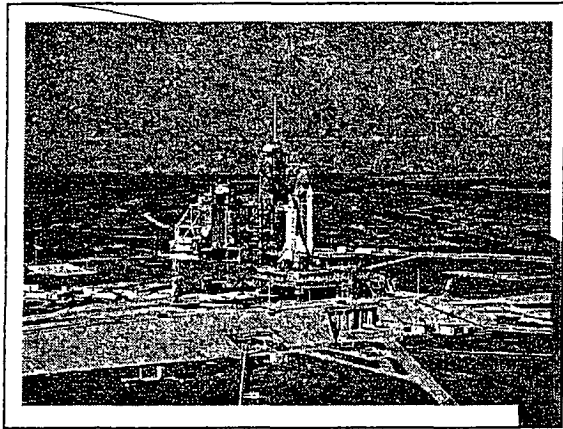
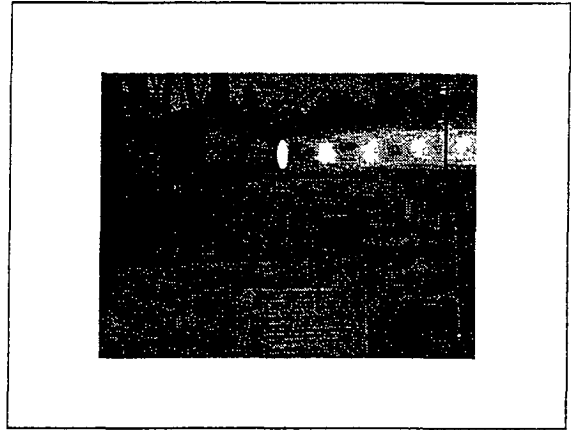
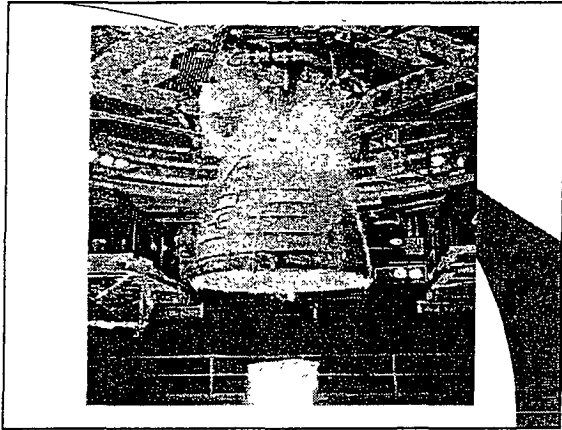
- Sound does not travel in vacuum, so it is not possible to talk in space. During EVA, Space walkers use radio
- Explosions occurring outside cannot be heard inside
- Talking inside Space Shuttle or ISS is same as on Earth; Air inside Shuttle is similar to ground
- N<sub>2</sub> = 80%, O<sub>2</sub> = 20%, Sea Level pressure = 14.7 psi similar to Earth

### Noise at and during Liftoff

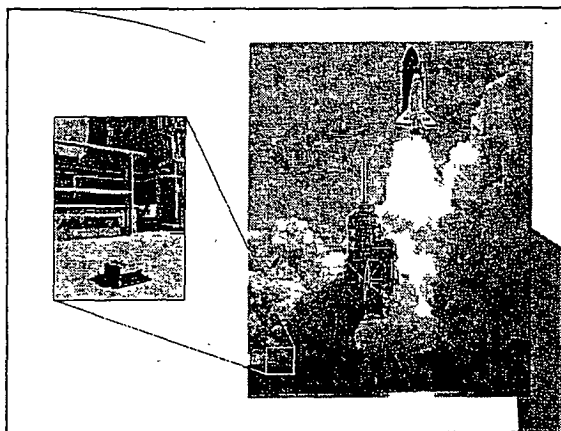
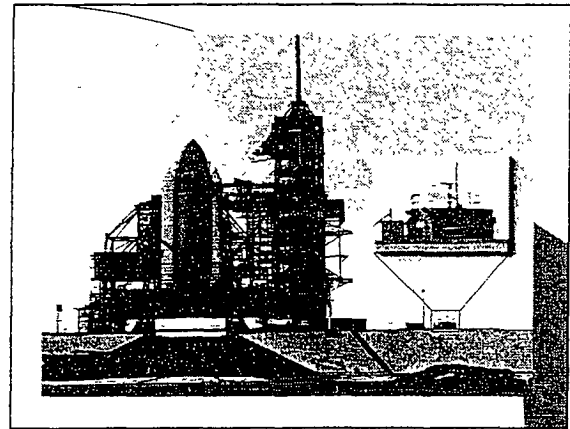
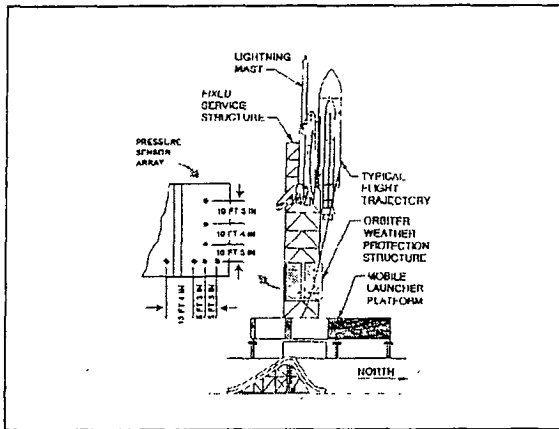
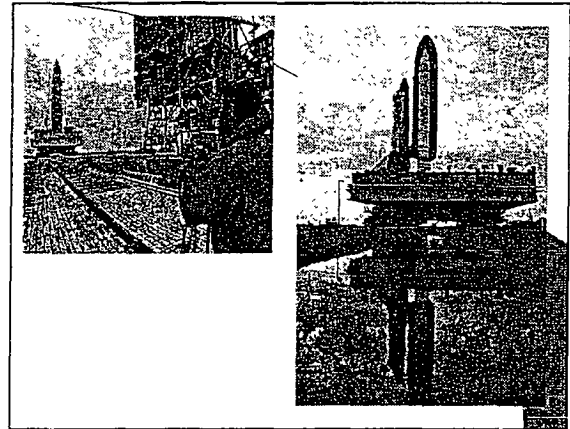
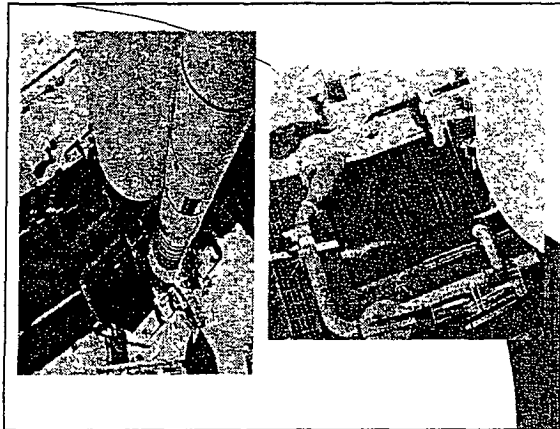
- Vibrations steadily increase up to Mach 1, the speed of sound, where shock waves add to the shaking; vibrations moderate as the Shuttle rises into thinner air as it ascends
- When SRB's burn out and are separated a dramatic change occurs; marked by a bang due to explosive bolts being fired; noise and vibration end at last

### Noise at and during Liftoff

- SRB separation occurs around 25 miles; air is very thin here for wind noise or shock waves to shake the Astronauts
- SRB separation occurs 2 minutes after liftoff (after T-0 seconds)
- Even though SSME's are still operating there is virtually no noise or vibrations during the rest of the ascent until External tank ejection (around 8.5 minutes after liftoff)





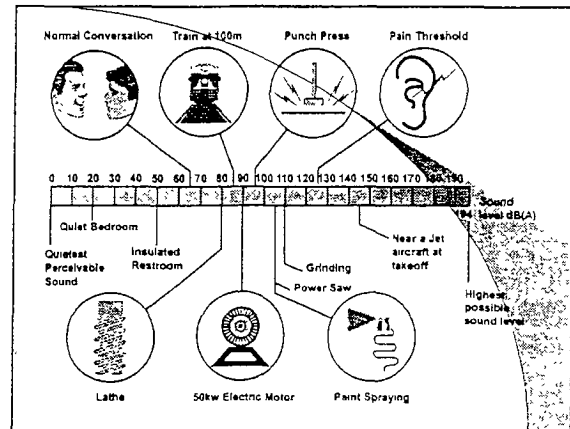


### Noise...Characterization

- Typical: Time history, OASPL, and FFT
- At NASA, we use Probability Density Distributions, Power Spectral Density, Cross-Power Spectral Density, Pressure Correlation Lengths, Coherences, etc.
- AJ factors – Vibroacoustic coupling between the sound field and structure of consequence

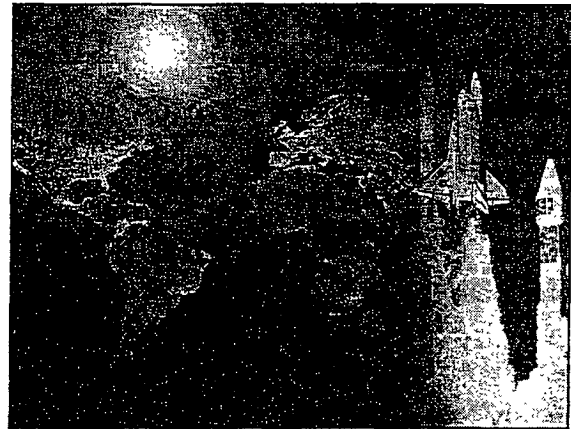
## Rocket vs. Common Noise

- 0 dB - Threshold of hearing
- 55 dB - Normal conversation
- 85 dB - Exposure must be < 8 hours
- 140 dB - Jet Engine (Pain threshold)
- 158 dB - Australian Cicada at 3 feet
- 180 dB - Shuttle lift-off at 10 feet
- 194 dB - Highest noise level (14.7 psi)

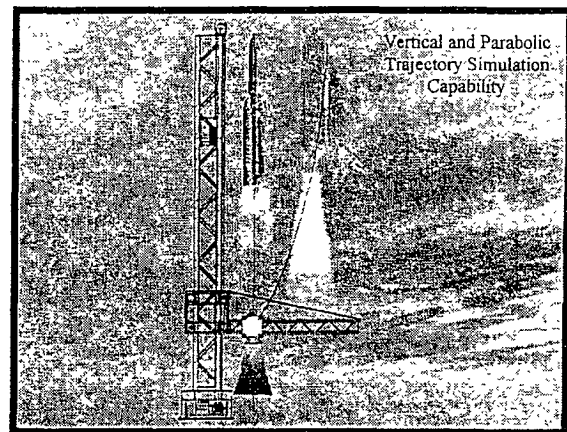
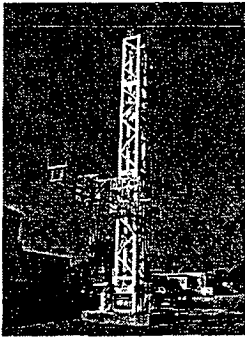


## Loudness : Records

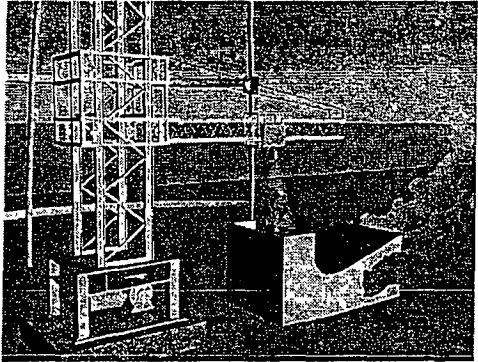
- 188 dB: Loudest Animal Sound
- 182 dB: Loudest Instrument
- 158 dB: Loudest Insect
- 145 dB: Loudest Animal
- 128 dB: Loudest Scream
- 122 dB: Loudest Whistle
- 121 dB: Loudest Shout
- 93 dB: Loudest Snoring



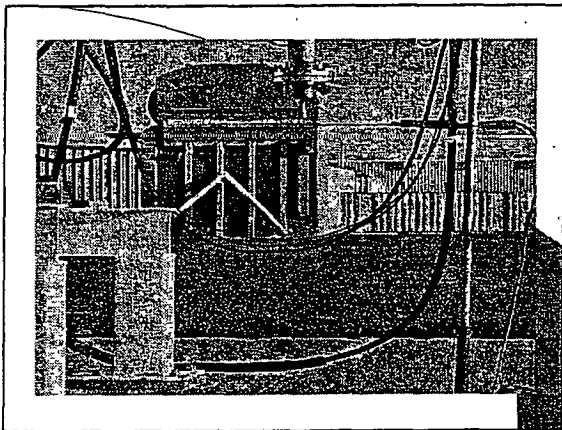
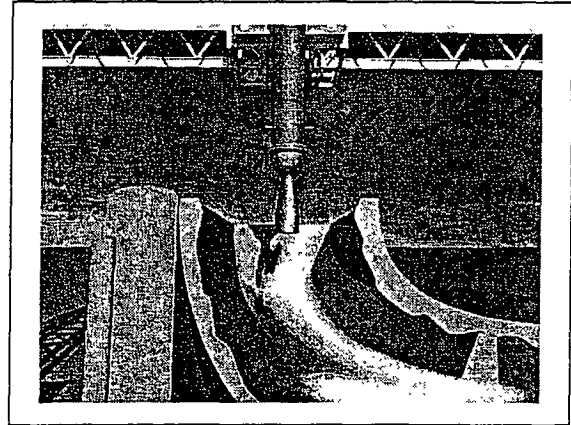
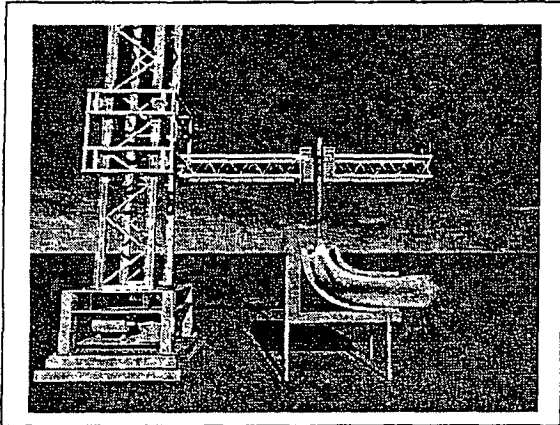
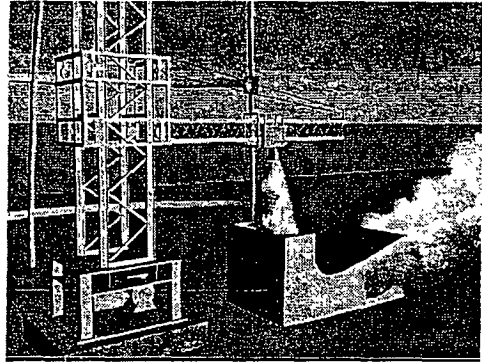
## Launch Systems Test bed



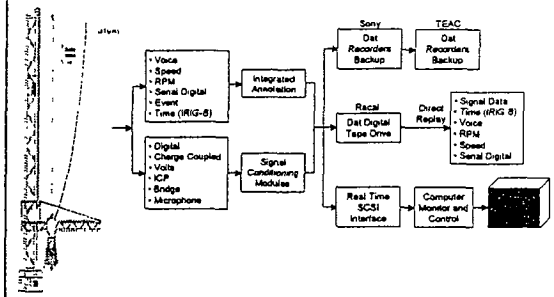
Cold Jet (GN2) Simulation with closed duct

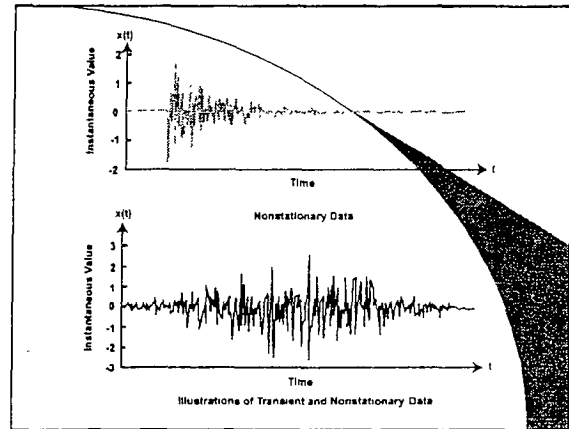
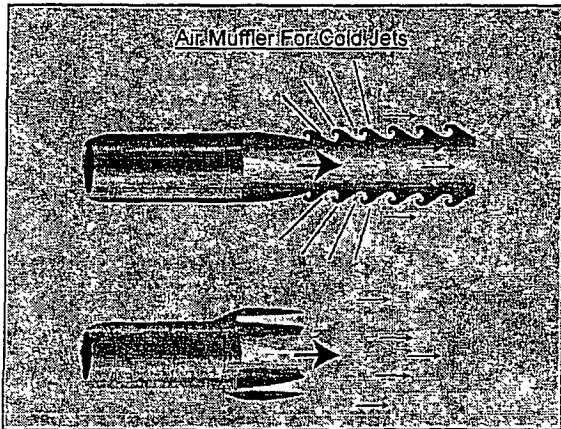


Future Hot Combusting Jet (GH2+GOX) Simulation



Trajectory Simulation Mechanism  
Data Acquisition System



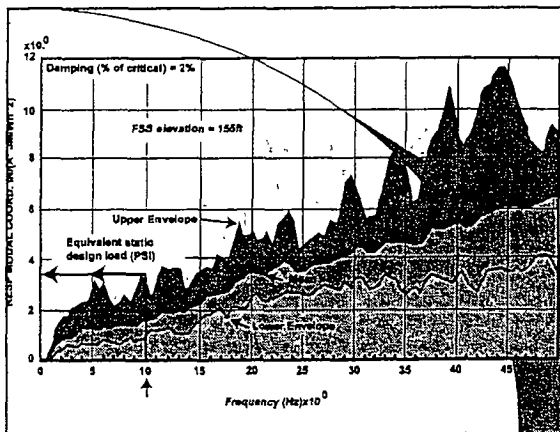


## Noise...Design Application

- What you don't hear...hurts you (20 Hz)
- Define acoustic load in terms of Static Equivalent Load (Architectural use)
- Dynamic Load: Enhance SEL analysis by conducting modal tests, response analysis; include considerations for stress, fatigue, and fracture mechanics

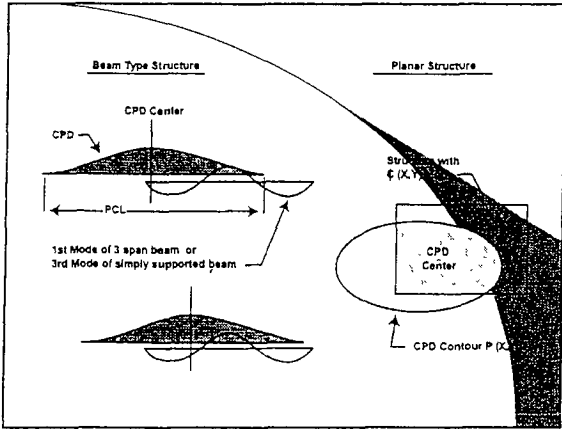
## Vibroacoustics Objectives

- Rocket Noise Models
  - Characterize acoustics environment
  - Bridge gap between test and analysis
  - Develop innovative analysis
- Vibration Response Analysis
  - Reliable Structural analysis method
  - Test project for verification
  - Implement system for use



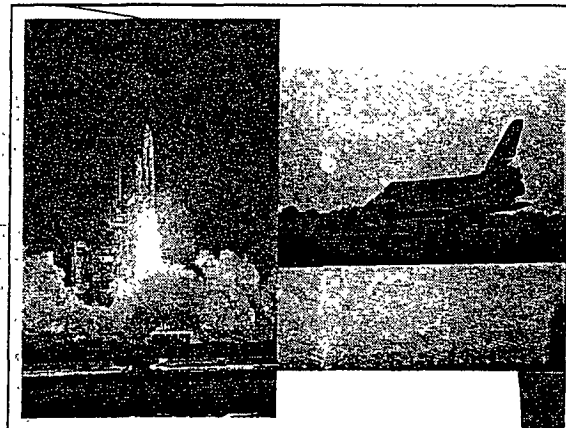
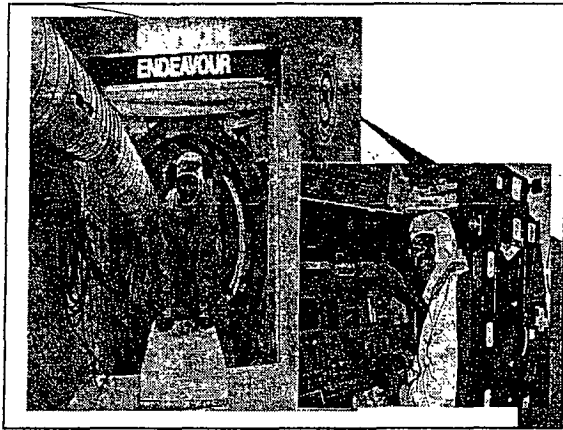
## Vibroacoustics...needs

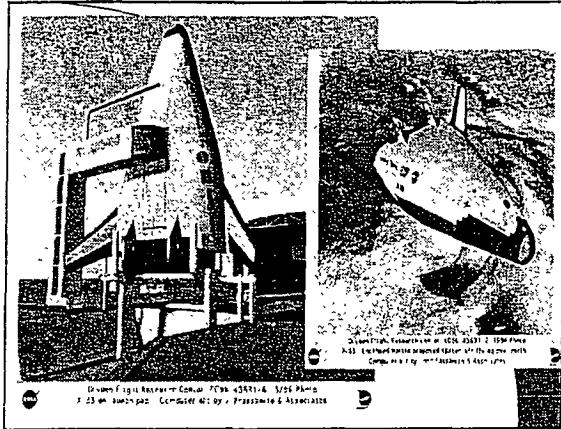
- ESL approach is crude; leads to over design – does not account for higher order modes
- Future launch vehicles have higher mechanical power – no experience
- Lower costs for new structures and modifications of older structures
- New measurement based analysis systems are needed coupled with FEA



## Response Analysis Methods

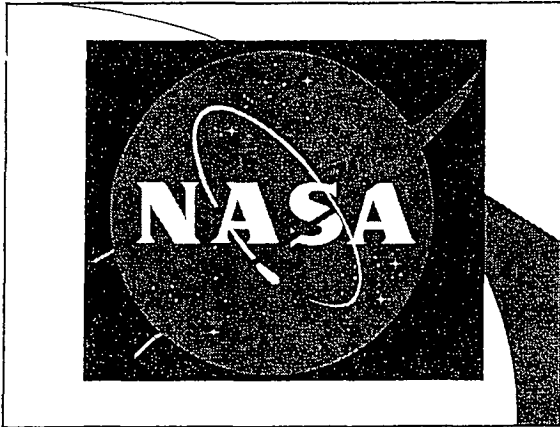
- Probabilistic Method (1967)
  - Based on classical random vibration
  - Transient – simulated as steady state
  - White Noise PSD, WN decay is CPD
- Deterministic Method (1987)
  - Based on non-stationary data (Shuttle)
  - Low frequency response analysis - key
  - Response Spectra and PCL, CPD, etc
  - True simulation of rocket response

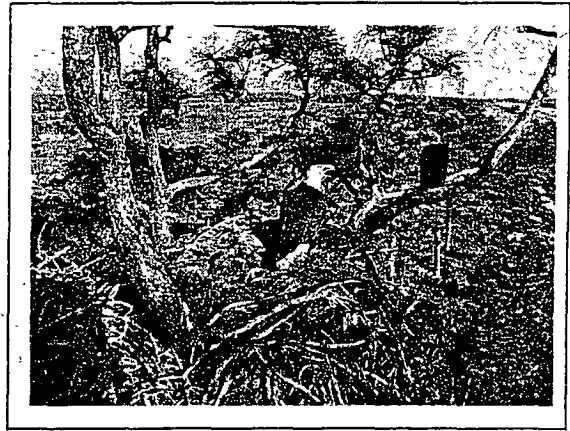
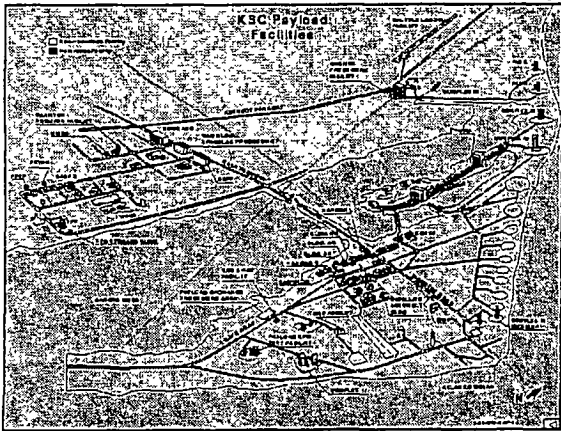
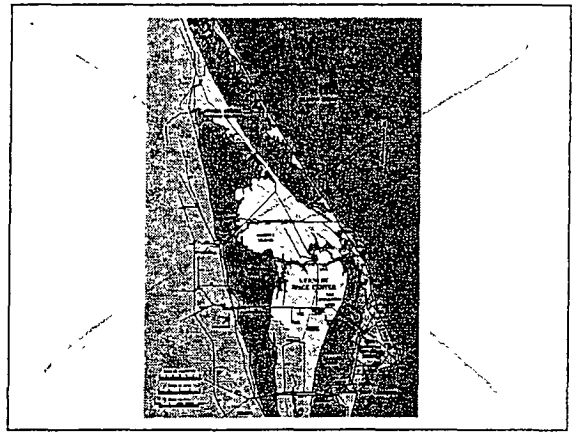
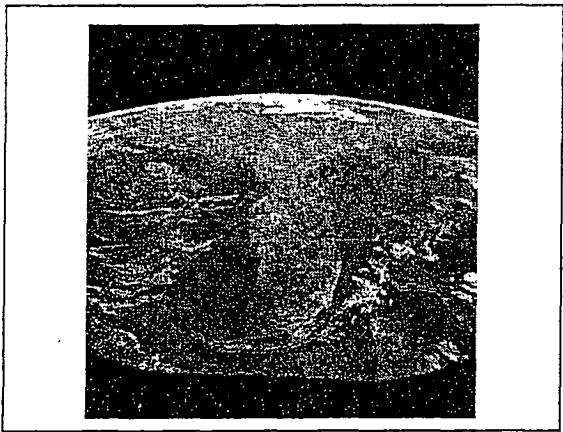
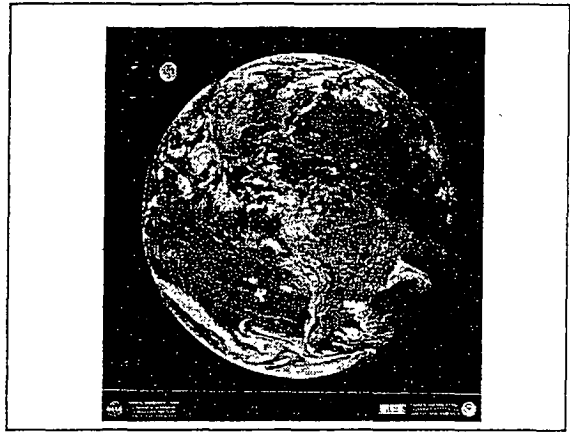
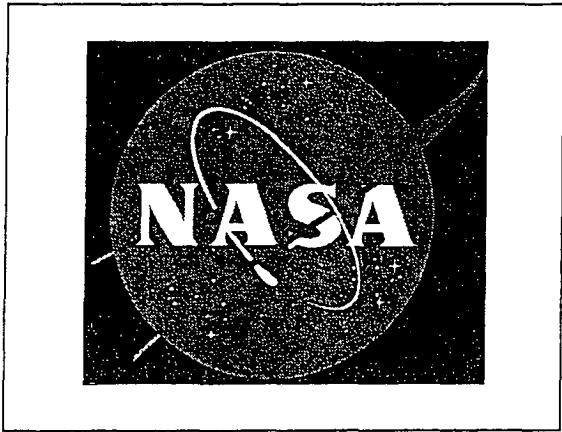


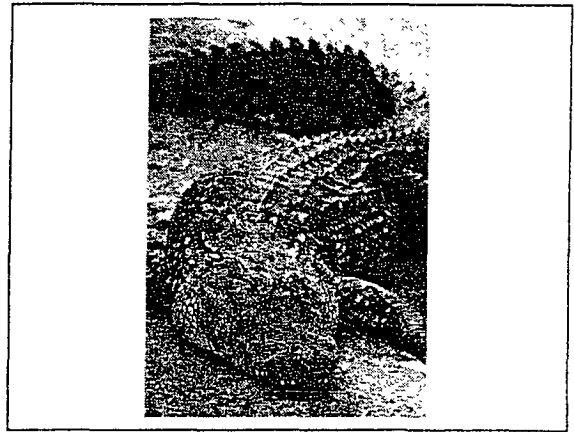
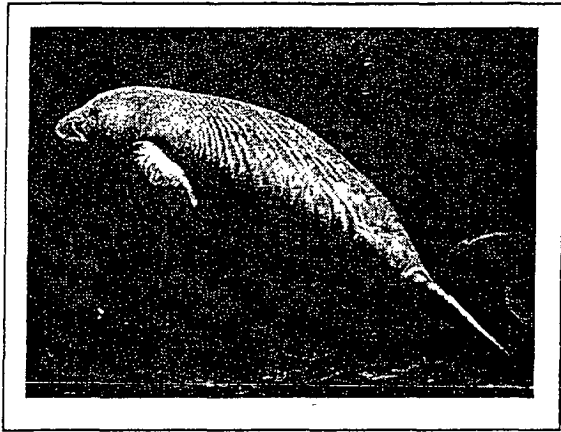


### Concluding Remarks

- Critical structures...bridges, oil rigs, aircrafts, rockets, and launch pads all require new techniques for design
- Dynamic loads..aerodynamic, earthquake, ocean waves, acoustic pressures...require random analysis
- Kobe earthquake....similar to water waves overlapping...cancel or additive
- Vibroacoustic coupling, launch exhaust management analysis....state of the art from a Ground Structures perspective





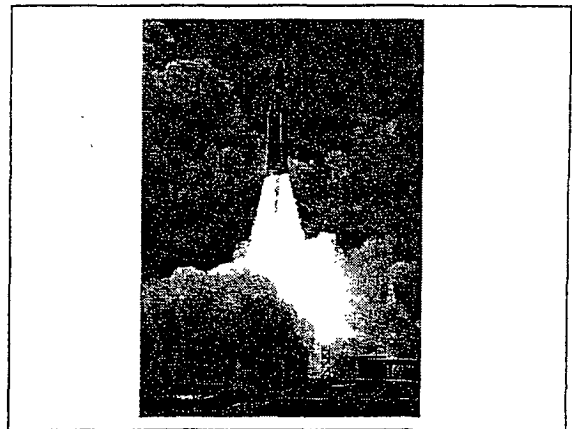
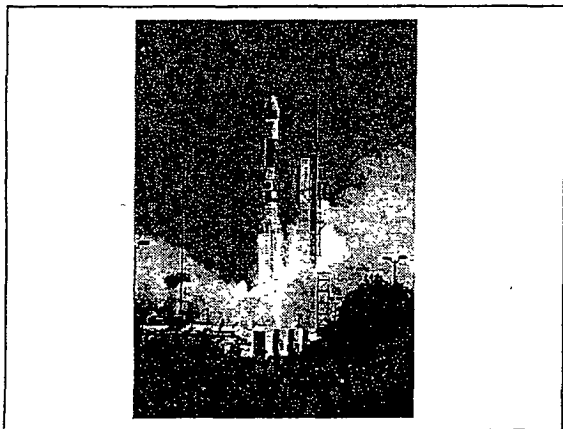
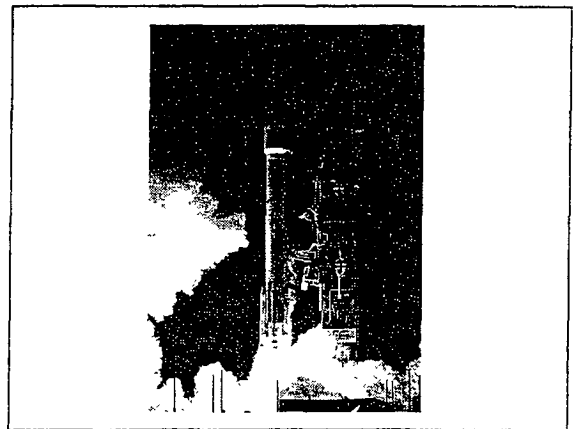


**Delivery to Launch**

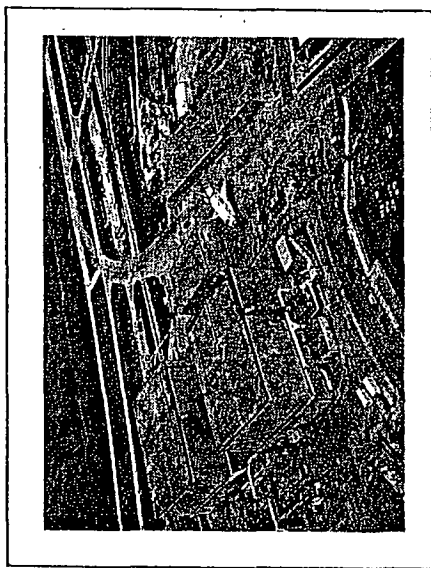
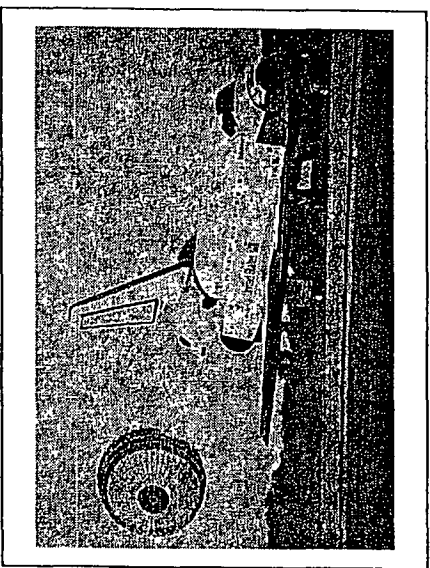
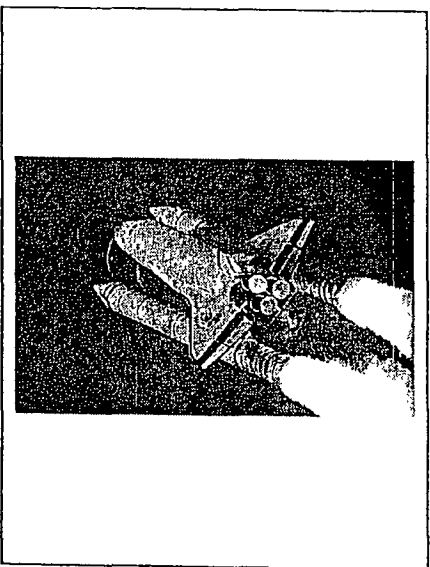
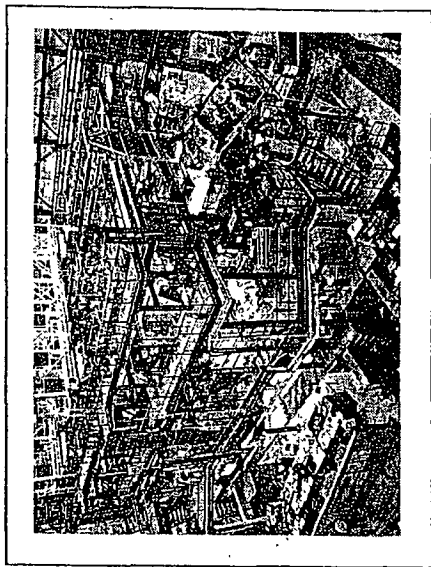
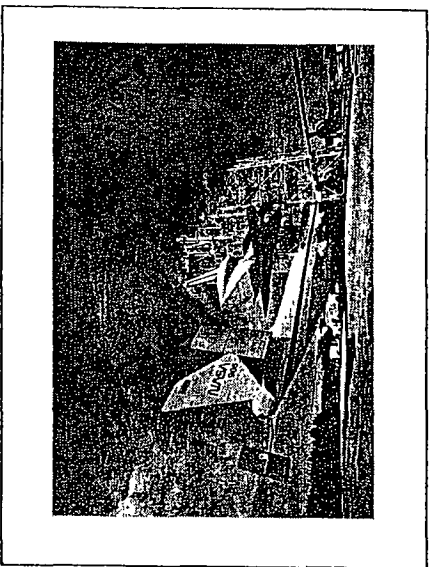
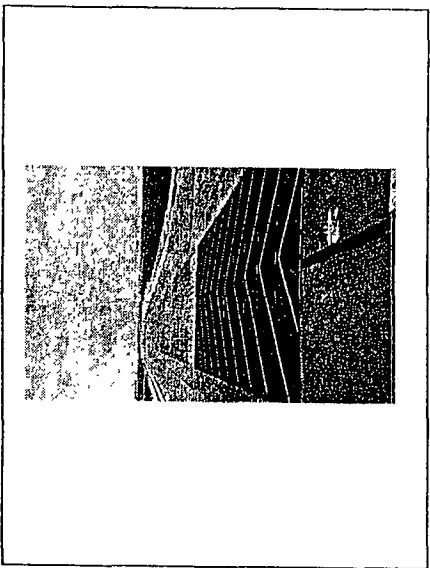
Hardware Delivery at KSC      Installation into Canister

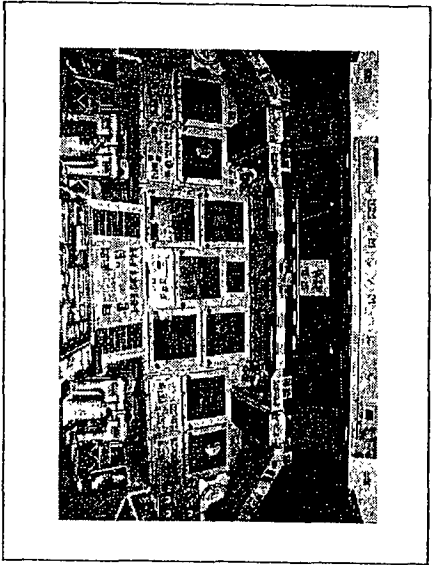
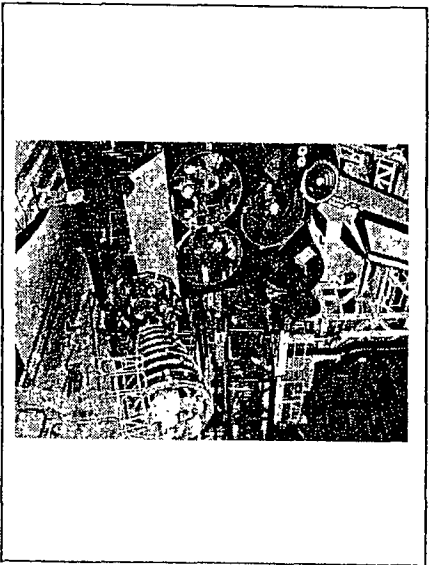
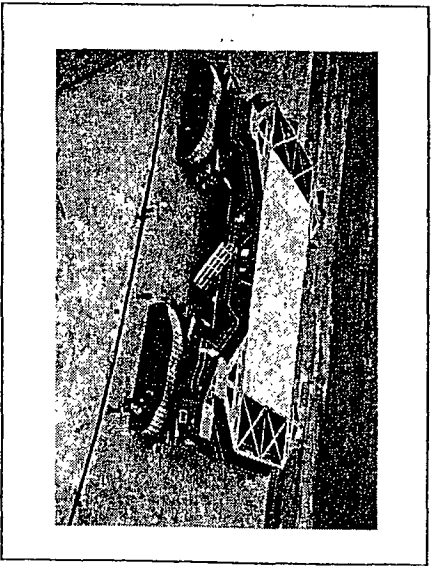
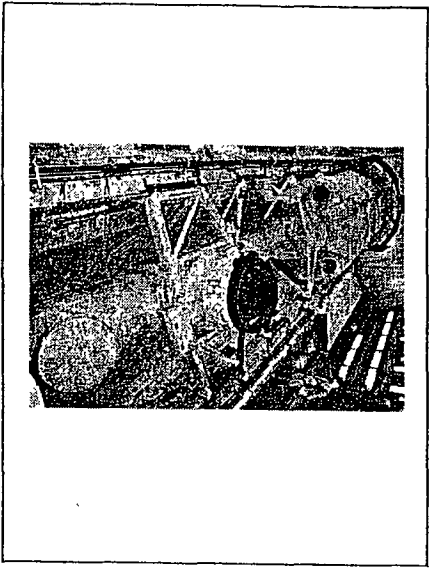
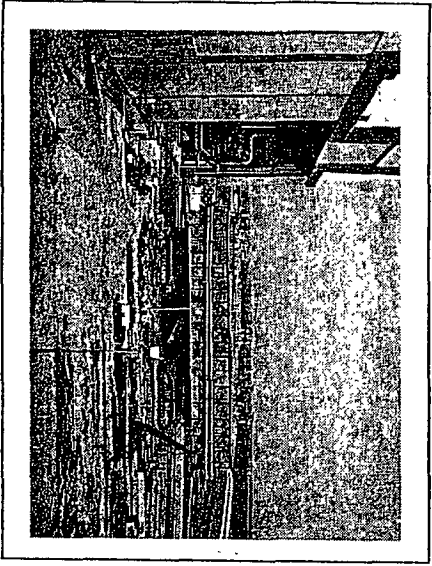
**Kennedy Space Center**

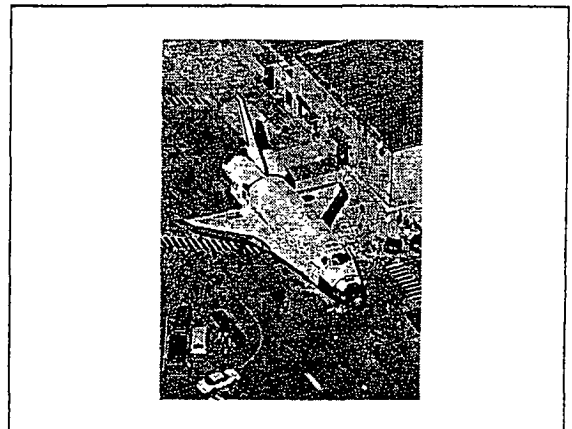
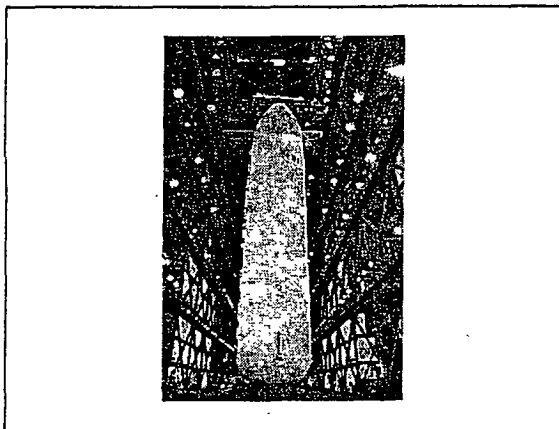
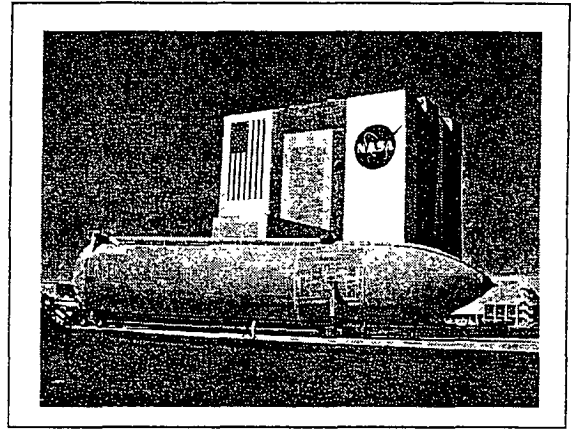
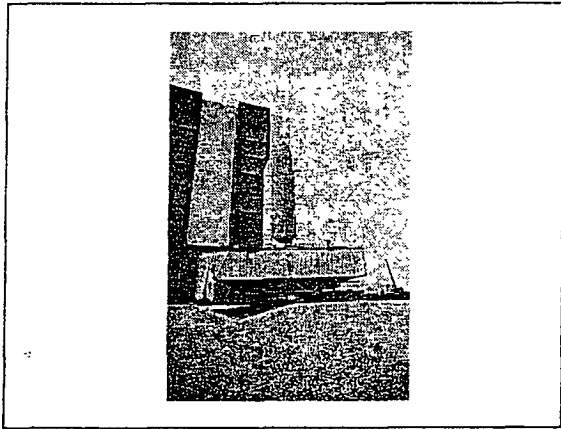
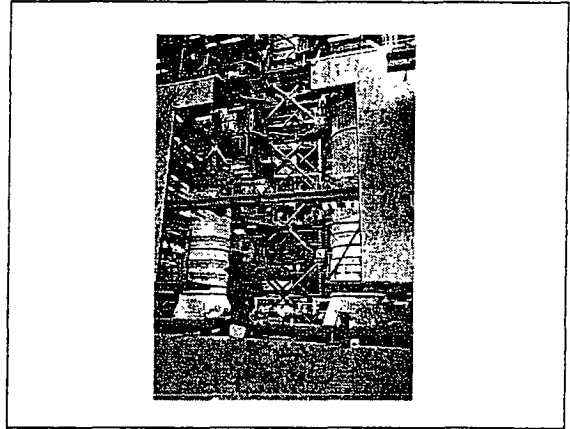
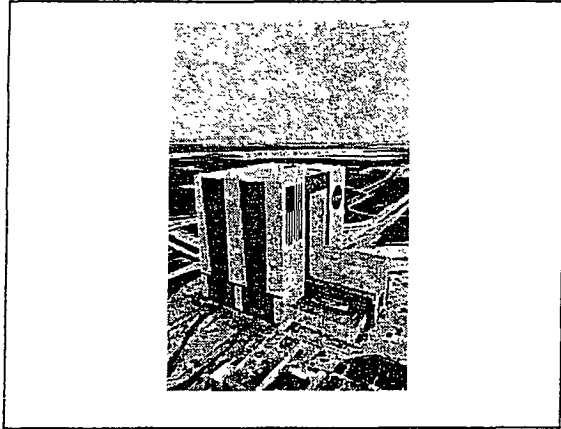
Payload Installation at Pad      Canister Rotation Facility

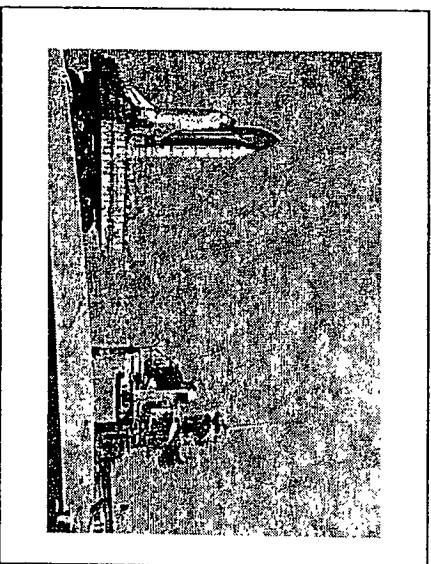
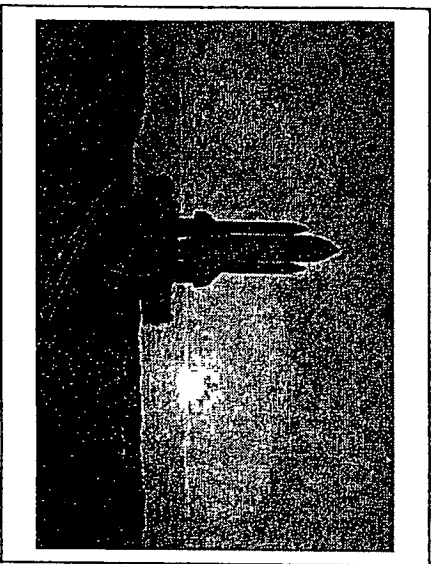
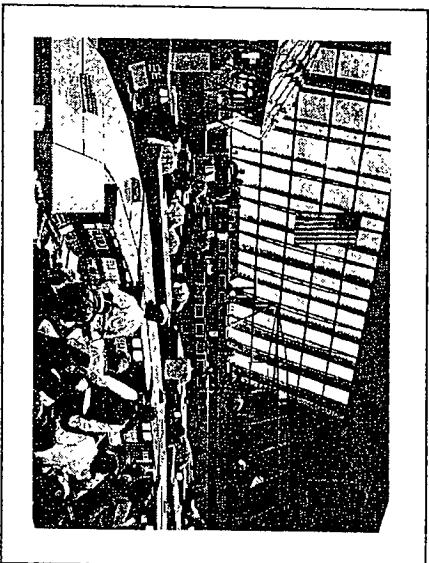
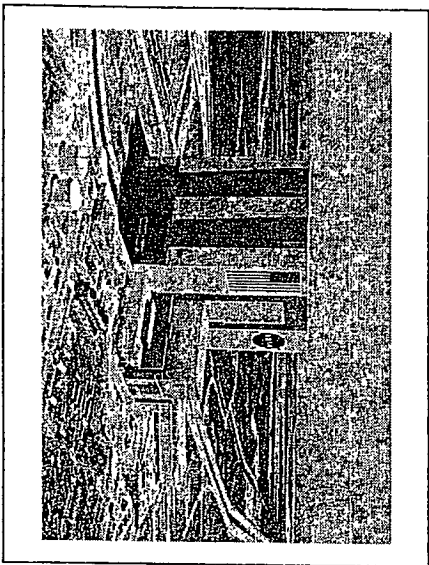
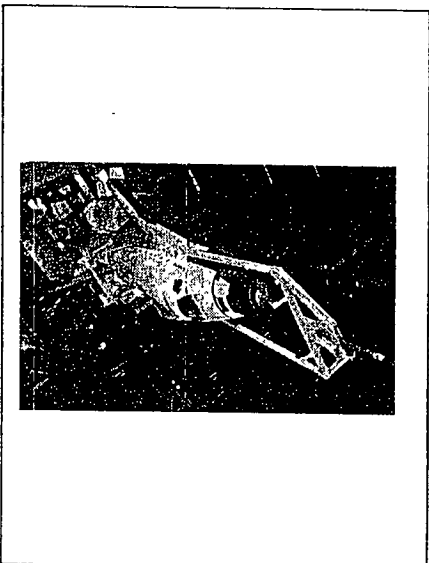
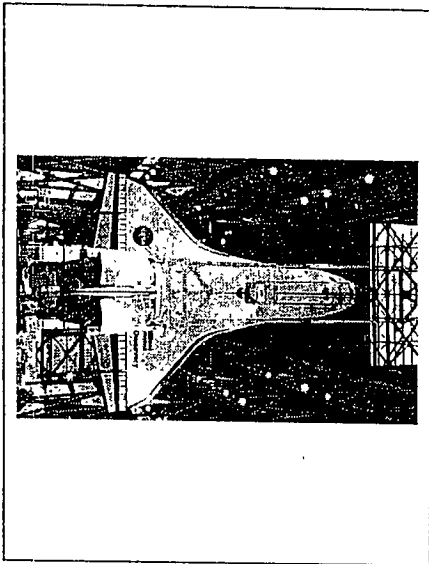
A collage of four small images showing the process of hardware delivery and installation at Kennedy Space Center. The top-left image shows hardware being delivered. The top-right image shows installation into a canister. The bottom-left image shows payload installation at the pad. The bottom-right image shows the canister rotation facility.

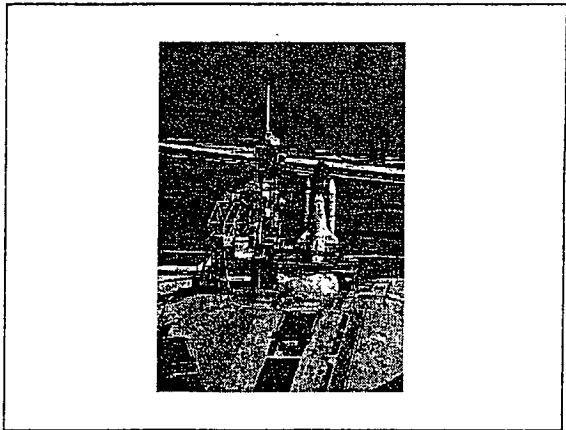
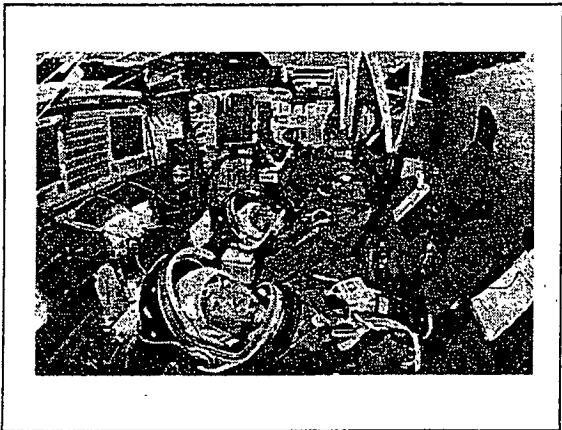
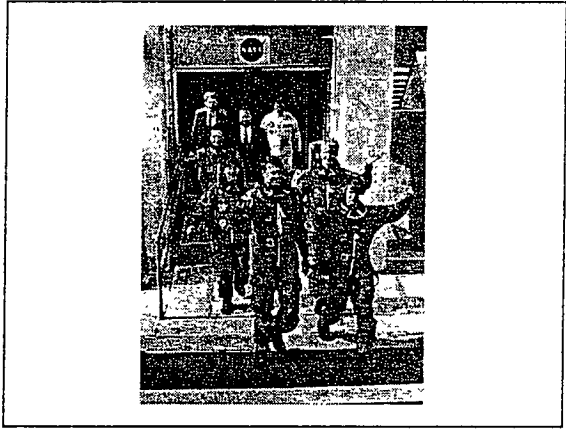
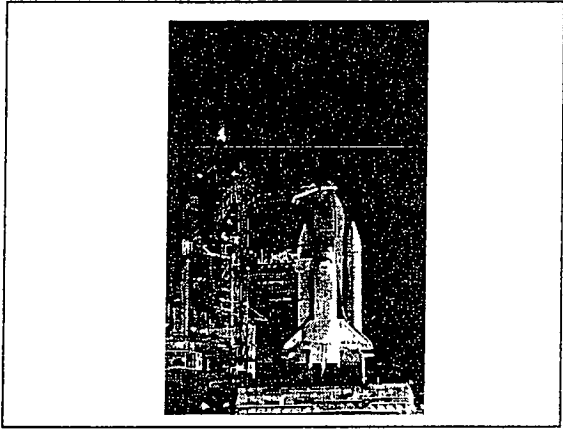
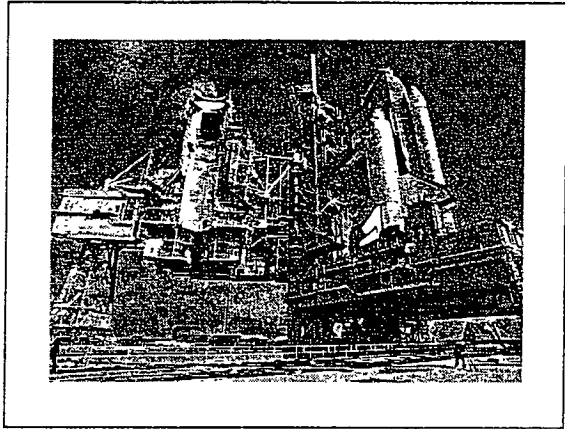
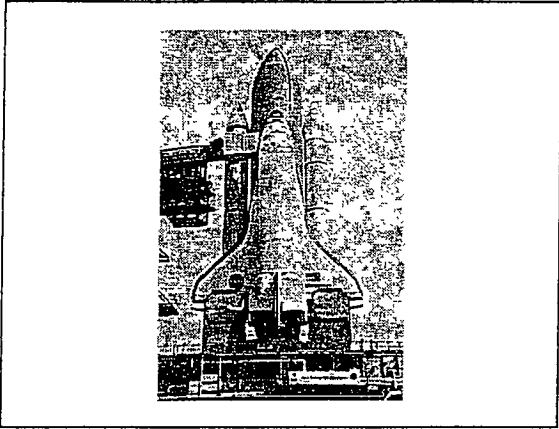


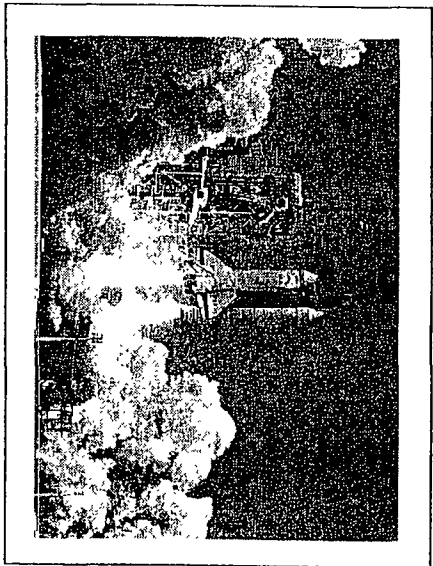
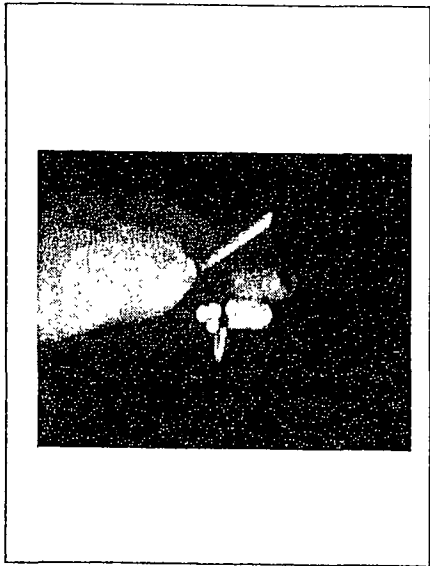
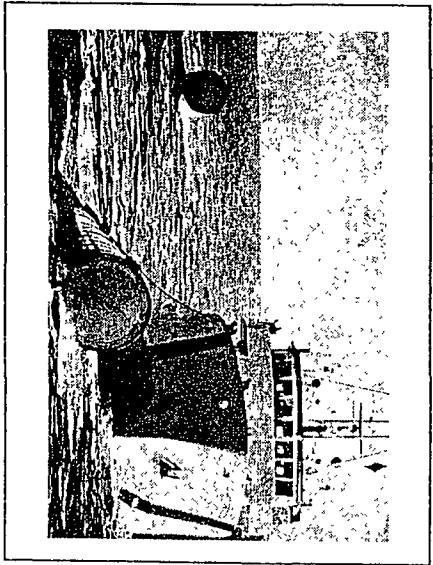
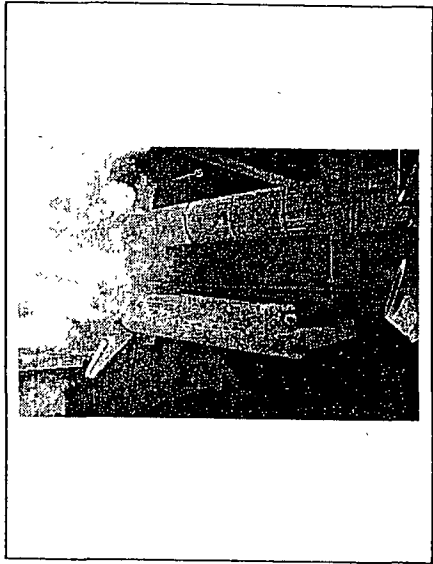
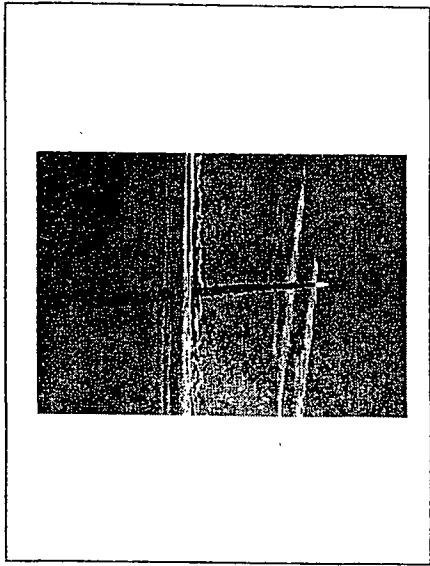
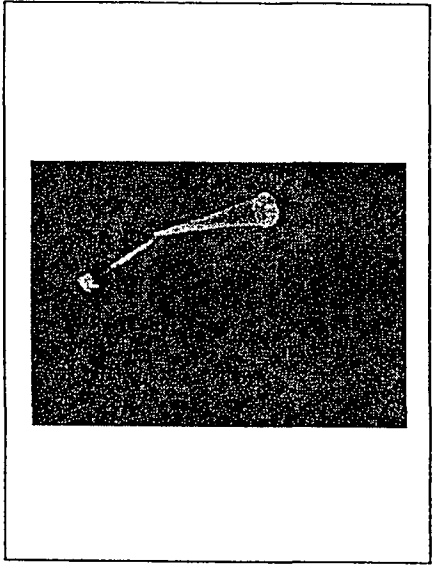


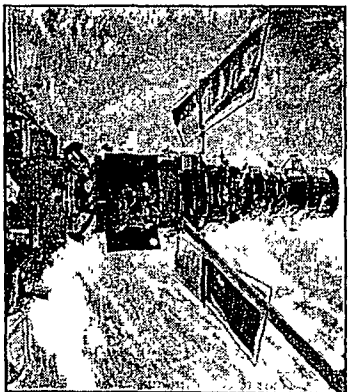
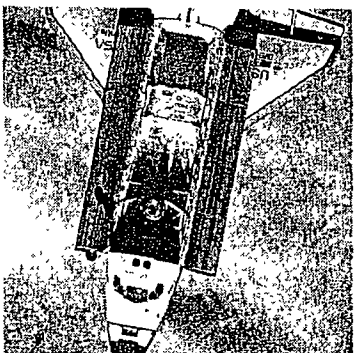
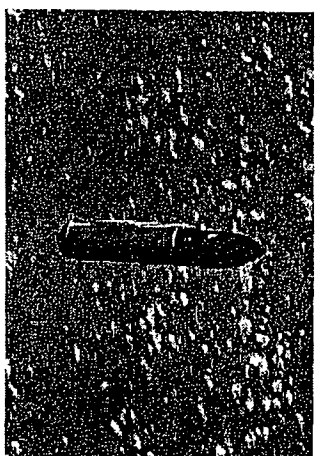
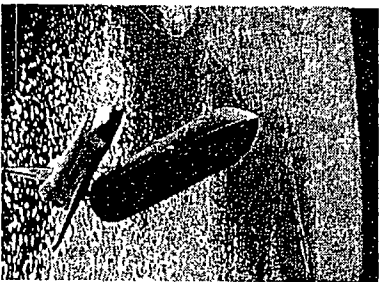












**Gasque's Pillars - M16**  
NASA Image ST S01-010-2, November 2, 1991.  
Astronauts and Science Officer (R) working on the exterior of the Space Shuttle Challenger.

**HST - WFPC2**

